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Designing a Frugal LEO Constellation with Integrated Orbital Micro Data Centers



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324 Satellites

LEO Constellation

\$400–450M

Total CAPEX

70–90%

Cost Reduction

25 TFLOPS

On-Orbit Compute

Executive Summary

Maargin is building a Space Network to provide communications services for users, enterprises, satellites, aircrafts, and other space entities.

This white paper presents the technical architecture behind Maargin’s low-cost LEO satellite constellation — designed to deliver broadband connectivity and on-orbit AI/ML compute for emerging markets at a **70–90% lower CAPEX** than existing mega-constellations.

Through five frugal design levers — higher-altitude 1,000–1,200 km orbits, 10–100 Gbps optical ISLs, SDR payloads, radiation-tolerant COTS electronics, and integrated 25 TFLOPS micro data centers — the 324-satellite constellation delivers a fully integrated connectivity and computing stack at **\$400–450M total CAPEX**.

The Problem

Mega-constellations cost \$5–10B+, locking out emerging markets. Over 2 billion people still lack reliable internet.

The Approach

Frugal engineering applied at constellation scale — higher altitude, COTS hardware, SDR payloads, orbital AI compute.

The Outcome

National broadband + orbital edge compute for India and partners at \$400–450M CAPEX, \$55–70M annual OPEX.

1 Constellation Design

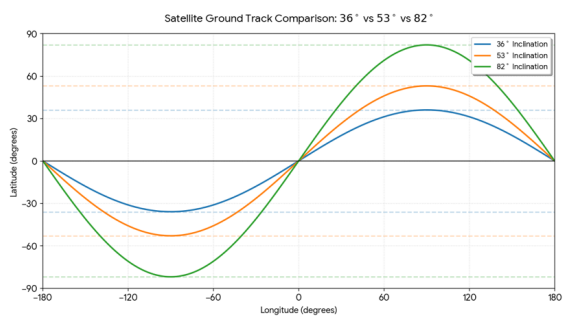
Orbital Architecture

Maargin’s three-shell Walker constellation at 1,000–1,200 km provides continuous coverage over India, UK/Europe, and polar regions from just 324 satellites.

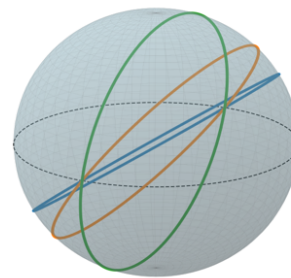
Shell	Region	Altitude	Inclination	Satellites
Shell 1	Tropical / India	1000–1200 km	20–40°	144
Shell 2	Temperate / Europe	1000–1200 km	53°	144
Shell 3	Polar	1000–1200 km	> 80°	36
Total				324

“Moving from 300 km to 1,200 km reduces the satellite count per shell by 89% — from 1,296 to 144 — cutting shell manufacturing cost from \$520M to \$60M.”

3D Visualization of Orbital Inclinations



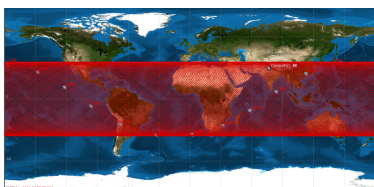
(a) Ground track



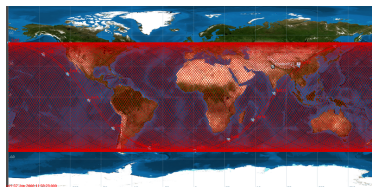
--- Equator — 36° Orbit — 53° Orbit — 82° Orbit

(b) Orbital inclination

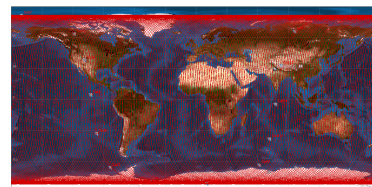
Figure 1: Orbital attributes for inclinations of 36°, 53°, and 82°.



(a) 36°



(b) 53°



(c) 82°

Figure 2: Ground tracks at various orbital inclinations.

Altitude	Satellites/Shell	Shell Cost	Launch Cost
300 km	1,296	\$520M	\$320M
600 km	484	\$200M	\$120M
1,200 km	144	\$60M	\$36M

2 Frugal Technology Stack

Optical Inter-Satellite Links

- 10–100 Gbps bidirectional
- Max range 4,000 km
- 1,550 nm wavelength
- 7–11 kg (vs. 15–25 kg standard)
- 40–50% weight reduction

Software-Defined Radio

- Development: weeks vs. 12–24 months
- NRE: minimal vs. \$1M per design
- In-orbit reconfigurable
- Ku-band + Ka-band capable
- Remote software updates

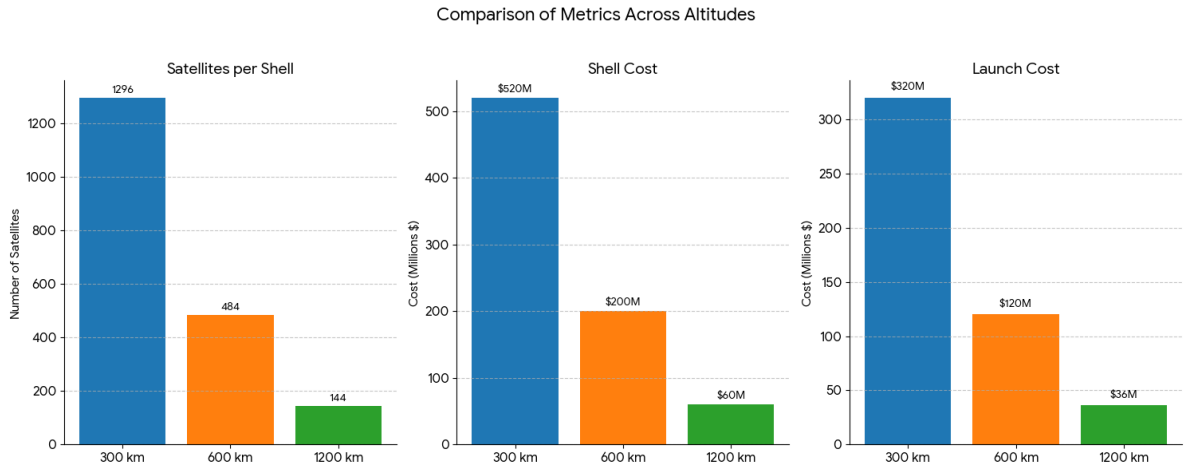


Figure 3: Satellite count, shell cost, and launch cost vs. altitude.

Radiation-Tolerant COTS

- Automotive/medical-grade COTS parts
- \$20–40k/sat vs. \$200k/sat
- 80–90% cost reduction
- TID below 10 krad(Si) with shielding
- ECC + watchdog fault management

Compact Satellite Bus

- 200–250 kg vs. 400+ kg standard
- Mfg. cost \$400–500k vs. \$1M
- 50% mass reduction
- 400 W (comms) / 600 W (with compute)
- 6-year design life

3 Orbital Micro Data Centers

Compute modules deployed on **≈108 satellites** — one-third of the constellation — forming a distributed orbital AI/ML fabric built atop the optical ISL network.

25 TFLOPS

Compute per node

2 TB

Persistent Storage

200 W

Avg. Power Draw

20–35 kg

Module Mass

108

Compute Satellites

Workload	Description
AI/ML Inference	Anomaly detection, workload management, traffic routing
Edge Processing	On-orbit data processing and results transmission
Comms Control	Autonomous routing, downlink prioritisation
Thermal Scheduling	Intensive compute during eclipse for peak radiative cooling

4 Economics

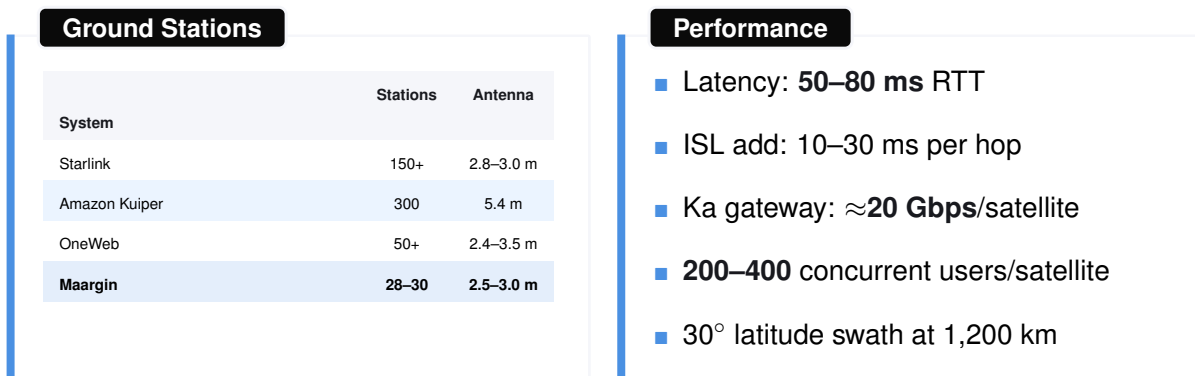


Cost Category	Amount (M\$)
Satellite design, testing & qualification	80–100
Satellite manufacturing (340 units @ \$400–500k)	136–170
Launch services (\$1,000–1,500/kg)	80–100
Licensing, spectrum, ITU, legal	5–10
Ground stations (30 stations + backhaul)	60–100
Total CAPEX	400–480

Parameter	State-of-the-Art LEO	Margin
Satellite mass	400+ kg	200–250 kg
Mfg. cost / satellite	\$1,000,000	\$300–400k
Shell deploy (manufacturing)	\$520M	\$60M
Shell deploy (launch)	\$320M	\$36M
Cost reduction	—	88–90%

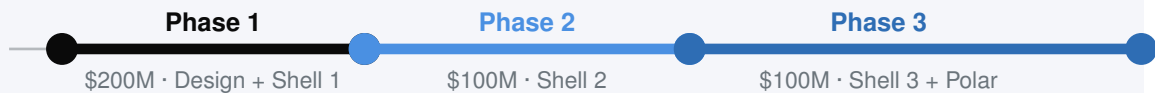
Annual OPEX post-deployment: **\$55–70M/year** (satellite replenishment, ground ops, backhaul, insurance).

5 Ground Segment & Performance



Frequency plan: Optical ISLs at 1,550 nm (10–100 Gbps) · Ku-band 10.7–14.5 GHz (user terminals) · Ka-band 17.8–30.0 GHz (gateways)

6 Deployment Roadmap



Phase 1 anchored by IN-SPACE and sovereign co-investment. Phases 2–3 open to private operators and regional partners as risk is retired and demand is validated — a template adaptable to any emerging space nation.

Conclusion

“Maargin proves that frugal engineering can deliver national-scale broadband + orbital AI compute for under \$500M — opening the space economy to the nations that need it most.”

Maargin’s frugal LEO constellation delivers **70–90% CAPEX savings** versus state-of-the-art mega-constellations while maintaining competitive performance. The \$400–450M total deployment cost makes sovereign space-based connectivity and orbital AI/ML compute accessible to emerging space nations.

The five-lever frugal architecture — orbit optimisation, SWaP-minimised payloads, COTS avionics with shielding, SDR-based software flexibility, and integrated orbital micro data centers — combined with a phased PPP deployment model provides a replicable template for any nation seeking sovereign LEO infrastructure.

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